

# Solution to

HW #3

MIM 1440

## Problem 4.1

Stiffness: Resistance to Deformation

Strength: Resistance to Failure

Ductility: Ability to Deform (Indication of How Deformable)

Yielding: Strength at the boundary of elastic Plastic Regime.

Toughness: Energy/Volume to break a Material

Strain Hardening: <sup>Indication of</sup> Material hardening with Deformation.

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## Problem 4.2

Brittle: No Plastic Deformation Prior to Failure

Ductile: Deform Plastically Prior to Failure

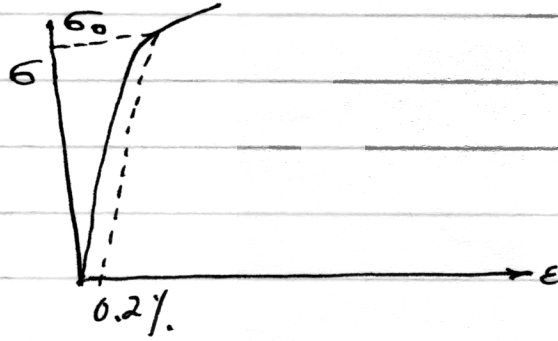
Resilient: Energy stored in the elastic Regime.

Tough: energy Required to break

Stiff: Resistance to Deformation: such spring constant

Strong: high stress for failure.

4.3



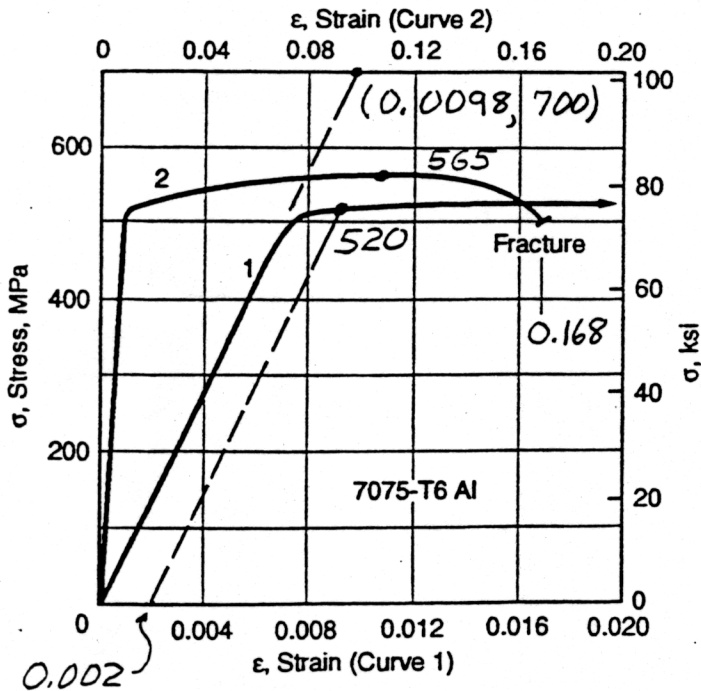
The offset  
is better, because  
it depends less

on the sensitivity of the instrument used.

offset may not be useful if you would like to avoid permanent deformation.

4.5

4.5

7075-T6 Al,  $d_i = 9.07$ ,  $d_f = 7.29$  mm

← 0.0098 from  
Curve 1 scale

← 0.168 from  
Curve 2 scale

$\sigma_f = 500$  MPa  
 $\epsilon_f = 0.168$

$$E = \frac{\Delta \sigma}{\Delta \epsilon} = \frac{700 \text{ MPa}}{0.0098} = 71,400 \text{ MPa} = 71.4 \text{ GPa} \blacktriangleleft$$

$$\text{At } \epsilon_{p0} = 0.002, \sigma_0 = 520 \text{ MPa} \blacktriangleleft$$

$$\sigma_u = 565 \text{ MPa} \blacktriangleleft$$

$$100 \epsilon_f = 16.8 \%, 100 \epsilon_{pf} \approx 16.1 \% \blacktriangleleft$$

$$\epsilon_{pf} \approx \epsilon_f - \frac{\sigma_f}{E} = 0.168 - \frac{500}{71,400} = 0.161$$

$$\% RA = 100 \frac{d_i^2 - d_f^2}{d_i^2} = 100 \frac{9.07^2 - 7.29^2}{9.07^2} = 35.4 \% \blacktriangleleft$$

Problem: 4.12

Engineering stress is force/original cross section =  $F/A_0$

$$\text{true stress} = \frac{F}{A} \rightarrow \text{cross section at any time.}$$

Engineering strain is  $\frac{L-L_0}{L_0}$  percentage based on the original length.

True strain is the summation of all incremental strain

$$\tilde{\epsilon} = \frac{\Delta L_1}{L_1} + \frac{\Delta L_2}{L_2} + \frac{\Delta L_3}{L_3} + \dots = \sum \frac{\Delta L_i}{L_i} = \int \frac{dL}{L} = \ln \frac{L}{L_0}$$

problem 4.15

4.15 Find  $\sigma_u$  in terms of  $H$  and  $n$ .

$$\tilde{\sigma}_u = H \tilde{\epsilon}_u^n, \quad n \approx \tilde{\epsilon}_u, \quad \tilde{\sigma}_u = H n^n$$

$$\sigma_u = \frac{\tilde{\sigma}_u}{1 + \epsilon_u} = \frac{\tilde{\sigma}_u}{e^{\tilde{\epsilon}_u}} = \frac{\tilde{\sigma}}{e^n} = H \left(\frac{n}{e}\right)^n$$

Note: Since  $\tilde{\epsilon}_u = \ln(1 + \epsilon_u)$ , we have

$$e^{\tilde{\epsilon}_u} = 1 + \epsilon_u$$

# Problem 4.21

$$E = 73.1 \text{ GPa} \quad \sigma_0 = 303 \text{ MPa} \quad \sigma_u = 476 \text{ MPa} \quad \text{Elongation}$$

= 20%. Reduction of Area 35%.

$$\tilde{\sigma}_B = 631 \text{ MPa} \quad \tilde{\epsilon}_f = 0.43 \quad H = 806 \quad n = 0.2$$

$$\tilde{\sigma} = H(\tilde{\epsilon})^n \Rightarrow \tilde{\sigma} = 806(\tilde{\epsilon})^{0.2} \quad \text{good up to necking}$$

$$\tilde{\epsilon} = 0.2$$

having true stress vs. true strain, engineering

stress vs. engineering strain will be

$$\tilde{\sigma} = \sigma(1 + \epsilon)$$

$$\tilde{\epsilon} = \ln(1 + \epsilon)$$

$$\tilde{\epsilon} = 0.2 = \ln(1 + \epsilon)$$

$$\epsilon = 0.221$$

$$\sigma(1 + \epsilon) = 806(\ln(1 + \epsilon))^{0.2}$$

$$\sigma = \frac{806(\ln(1 + \epsilon))^{0.2}}{1 + \epsilon}$$

$$\frac{\sigma_0}{E} \leq \epsilon \leq 0.221$$

You can plot this  
Curve

$$4.14 \times 10^{-3} \leq \epsilon \leq 0.221$$

# Results

In the elastic Regime  $\sigma = E \epsilon$   $\epsilon \leq 4.14 \times 10^{-3}$

In the plastic Regime up to necking

$$\sigma = \frac{806 (\ln(1+\epsilon))^{0.2}}{1+\epsilon}$$

$$4.14 \times 10^{-3} \leq \epsilon \leq 0.221$$

$$\tilde{\sigma} = 806 (\tilde{\epsilon})^{0.2} \quad 0 \leq \tilde{\epsilon} < 0.2$$

Assuming Bridgman constant holds for Aluminum.

$$\tilde{\sigma}_B = B \tilde{\sigma}$$

$$B = 0.83 - 0.186 \log \tilde{\epsilon}$$

$$\tilde{\sigma}_B = H (\tilde{\epsilon})^{0.2}$$

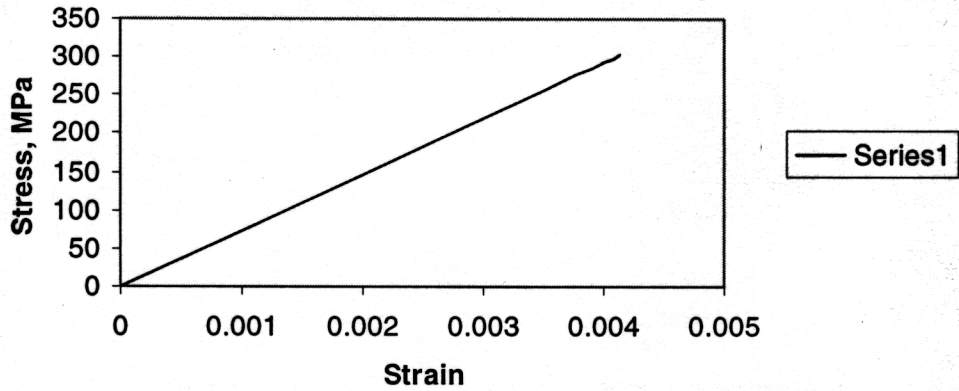
$$\tilde{\sigma} = \frac{H (\tilde{\epsilon})^{0.2}}{(0.83 - 0.186 \log \tilde{\epsilon})} \quad 0.2 \leq \tilde{\epsilon} < 0.43$$

$$\tilde{\sigma} = \sigma e^{\tilde{\epsilon}} \Rightarrow \sigma = \frac{\tilde{\sigma}}{e^{\tilde{\epsilon}}}$$

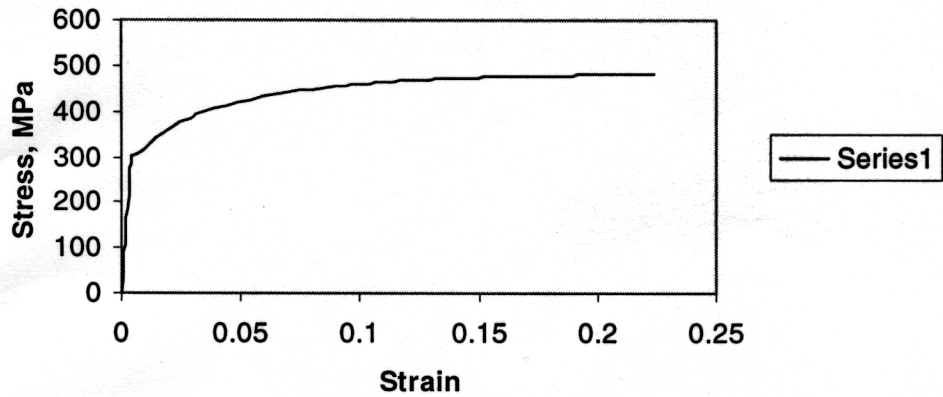
$$\sigma = \left( \frac{806 (\tilde{\epsilon})^{0.2}}{(0.83 - 0.186 \log \tilde{\epsilon})} \right) / e^{\tilde{\epsilon}}$$

$$0.2 < \tilde{\epsilon} < 0.43$$

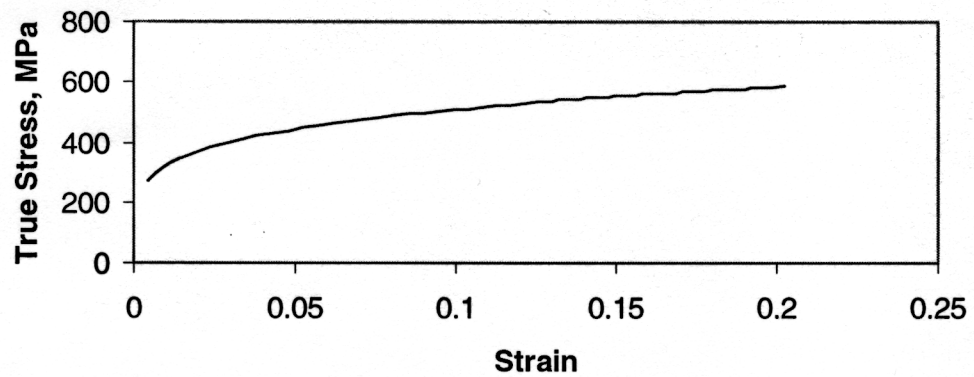
**Elastic Portion of Data**



**Engineering Stress Vs. Eng. Strain up to necking**



**True Stress vs. True Strain up to necking**



# True stress vs. true strain beyond necking

